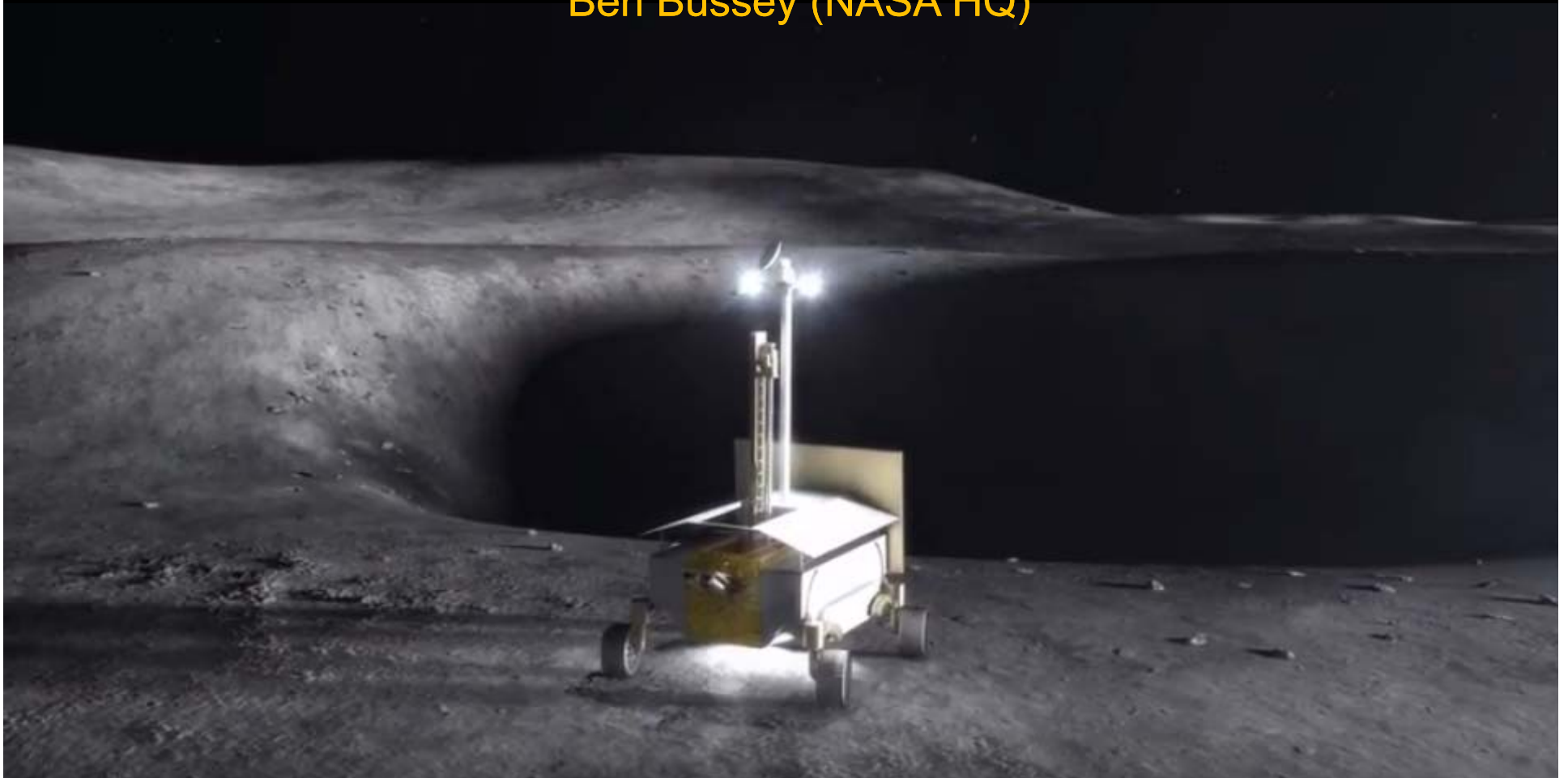


National Aeronautics and Space Administration



# Resource Prospector: Goals and Measurements

Anthony Colaprete (NASA ARC)  
Ben Bussey (NASA HQ)



# EVOLVABLE MARS CAMPAIGN

*A Pioneering Approach to Exploration*



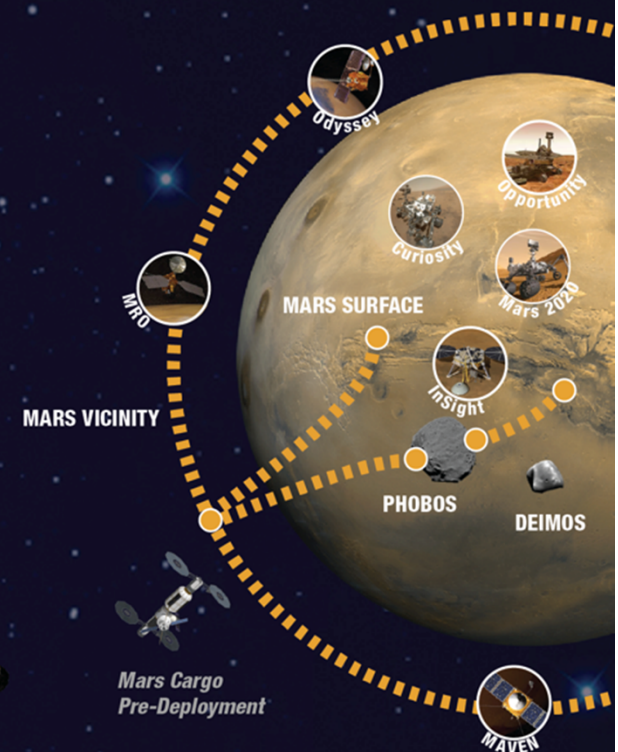
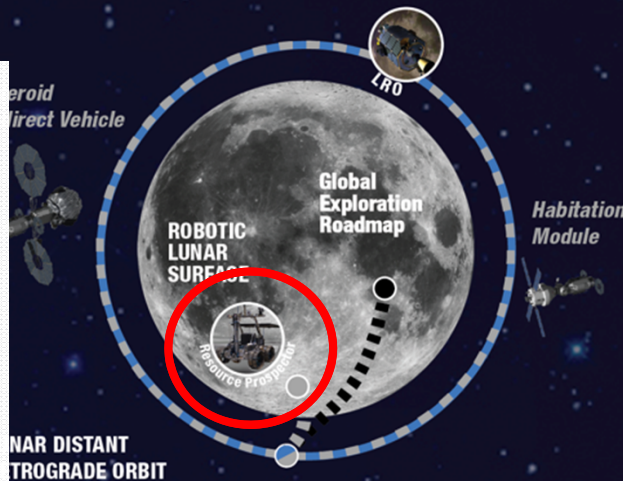
EARTH RELIANT

PROVING GROUND

EARTH INDEPENDENT

## RP Project Objectives:

Moon as a Resource & Proving Ground for Earth Independent ISRU



Commercial Cargo & Crew

Space Launch System 70 mt

Space Launch System 130 mt

## THE TRADE SPACE

Across the Board

Solar Electric Propulsion • In-Situ Resource Utilization (ISRU) • Robotic Precursors • Human/Robotic Interactions • Partnership Coordination • Exploration and Science Activities

Cis-lunar Trades

- Deep-space testing and autonomous operations
- Extensibility to Mars
- Mars system staging/refurbishment point and trajectory analyses

Mars Vicinity Trades

- Split versus monolithic habitat
- Cargo pre-deployment
- Mars Phobos/Deimos activities
- Entry descent and landing concepts
- Transportation technologies/trajectory analyses



# RP Mission Goals and Relevance



## From *LEAG Robotic Campaign Analysis (2011)*:



### Phase I: Lunar Resource Prospecting

- Defining the composition, form, and extent of the resource;
- Characterizing the environment in which the resources are found;
- Defining the accessibility/extractability of the resources;
- Quantifying the geotechnical properties of the lunar regolith in the areas where resources are found;
- Being able to traverse several kilometers and sample and determine lateral and vertical distribution on meter scales;
- Identifying resource-rich sites for targeting future missions

**Resource Prospector is aligned with the community vision for the next lunar resource mission**

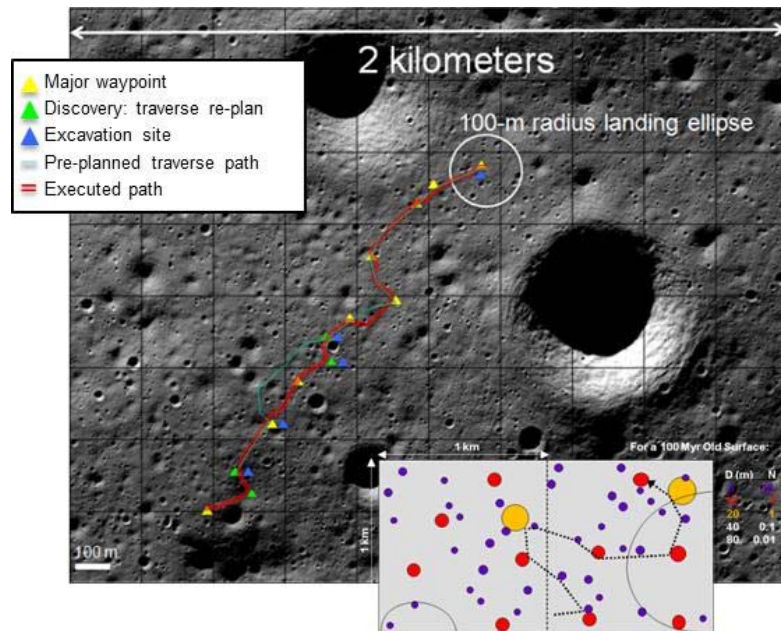


# Determining 'Operationally Useful' Deposits

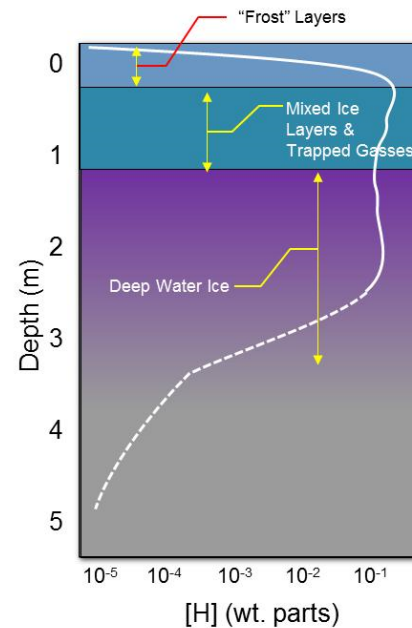


Need to assess the extent of the resource 'ore body'

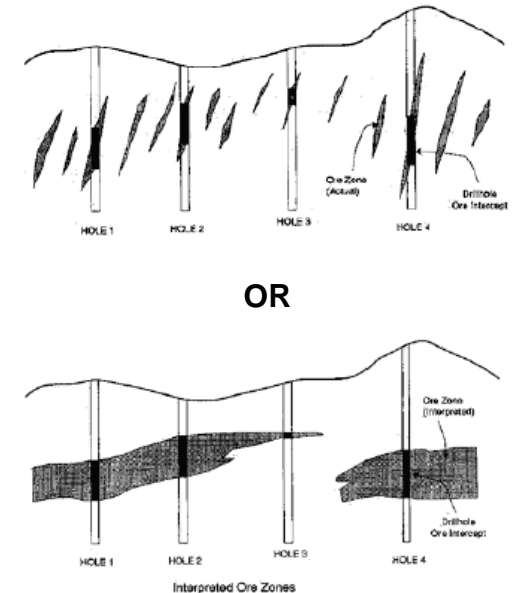
Need to Evaluate Local Region (1 to 3 km)



Need to Determine Vertical Profile



Need to Determine Distribution



An 'Operationally Useful' Resource Depends on What is needed, How much is needed, and How often it is needed

## Potential Lunar Resource Needs\*

- 1,000 kg oxygen ( $O_2$ ) per year for life support backup (crew of 4)
- 3,000 kg of  $O_2$  per lunar ascent module launch from surface to  $L_1/L_2$
- 16,000 kg of  $O_2$  per reusable lunar lander ascent/descent vehicle to  $L_1/L_2$  (fuel from Earth)
- 30,000 kg of  $O_2$ /Hydrogen ( $H_2$ ) per reusable lunar lander to  $L_1/L_2$  (no Earth fuel needed)

\*Note: ISRU production numbers are only 1<sup>st</sup> order estimates for 4000 kg payload to/from lunar surface

# RP Level 1 Requirements



## 1.1 RESOURCE PROSPECTOR SHALL LAND AT A LUNAR POLAR REGION TO ENABLE PROSPECTING FOR VOLATILES

- **Full Success Criteria:** Land at a polar location that maximizes the combined potential for obtaining a high volatile (hydrogen) concentration signature and mission duration within traverse capabilities
- **Minimum Success Criteria:** Land at a polar location that maximizes the potential for obtaining a high volatile (hydrogen) concentration signature

## 1.2 RESOURCE PROSPECTOR SHALL BE CAPABLE OF OBTAINING KNOWLEDGE ABOUT THE LUNAR SURFACE AND SUBSURFACE VOLATILES AND MATERIALS

- **Full Success Criteria:** Take **both** *sub-surface measurements of volatile constituents via excavation and processing* **and** *surface measurements*, at multiple locations
- **Minimum Success Criteria:** Take **either** *sub-surface measurements of volatile constituents via excavation and processing* **or** *surface measurements*, at multiple locations

# Measurement Requirement Summary



## Paraphrased Level 2 Measurement Requirements

### Minimum Success:

- Make measurements from two places separated by at least 100 meters
- Surface or subsurface measurements

### Full Success (shalls):

- Measurements from two places separated by at least 1000 meters
- Surface and subsurface measurements
- Measurements in and sample acquired from shadowed area
- Demonstrate ISRU

### Stretch Goals (shoulds):

- Make subsurface measurements in at least eight (8) locations across 1000 m (point-to-point) distance
- Process and analyze subsurface material in at least four (4) locations across 1000 m (point-to-point) distance
- Provide geologic and thermal context



# SKGs and RP – Address at Least 22 Lunar SKGs

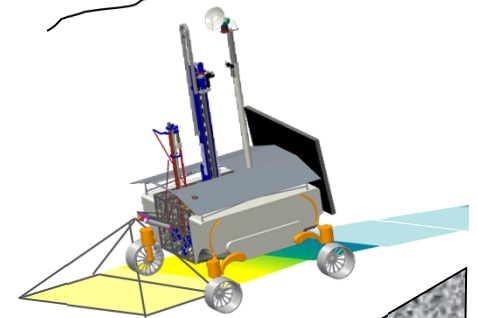


Lunar Exploration Strategic Knowledge Gaps			Instrument or Activity	RP Relevance
<b>I. Understand the Lunar Resource Potential</b>				
D-3	Geotechnical characteristics of cold traps		NIRVSS, Drill, Rover	H
D-4	Physiography and accessibility of cold traps		Rover-PSR traverses, Drill, Cameras	VH
D-6	Earth visibility timing and extent		Mission Planning	VH
D-7	Concentration of water and other volatiles species within depth of 1-2 m		NSS, NIRVSS, OVEN-LAVA	VH
D-8	Variability of water concentration on scales of 10's of meters		NSS, NIRVSS, OVEN-LAVA	VH
D-9	Mineralogical, elemental, molecular, isotopic, make up of volatiles		NIRVSS, OVEN-LAVA	VH- Volatiles LM-Minerals
D-10	Physical nature of volatile species (e.g. pure concentrations, intergranular, globular)		NIRVSS, OVEN-LAVA	H
D-11	Spatial and temporal distribution of OH and H <sub>2</sub> O at high latitudes		NIRVSS, OVEN-LAVA	M-H
D-13	Monitor and model movement towards and retention in PSR		NIRVSS, OVEN-LAVA	M
G	Lunar ISRU production efficiency 2		Drill, OVEN-LAVA, LAVA-WDD	M
<b>III. Understand how to work and live on the lunar surface</b>				
A-1	Technology for excavation of lunar resources		Drill, Rover	M
B-2	Lunar Topography Data		Planning Products, Cameras	M
B-3	Autonomous surface navigation		Traverse Planning, Rover	M-L
C-1	Lunar surface trafficability: Modeling & Earth Tests		Planning, Earth Testing	M
C-2	Lunar surface trafficability: In-situ measurements		Rover, Drill	H
D-1	Lunar dust remediation		Rover, NIRVSS, OVEN	M
D-2	Regolith adhesion to human systems and associated mechanical degradation		Rover, NIRVSS, OVEN, Cameras	M
D-3	Descent/ascent engine blast ejecta velocity, departure angle, and entrainment mechanism: Modeling		Landing Site Planning, Testing	M
D-4	Descent/ascent engine blast ejecta velocity, departure angle, and entrainment mechanism		Lander, Rover, NIRVSS	H
F-2	Energy Storage - Polar missions		Stretch Goal: Lander, Rover	
F-4	Power Generation - Polar missions		Rover	M

# Prospecting...



1. While roving, prospecting instruments search for enhanced surface  $\text{H}_2\text{O}/\text{OH}$ , other volatiles and volumetric hydrogen

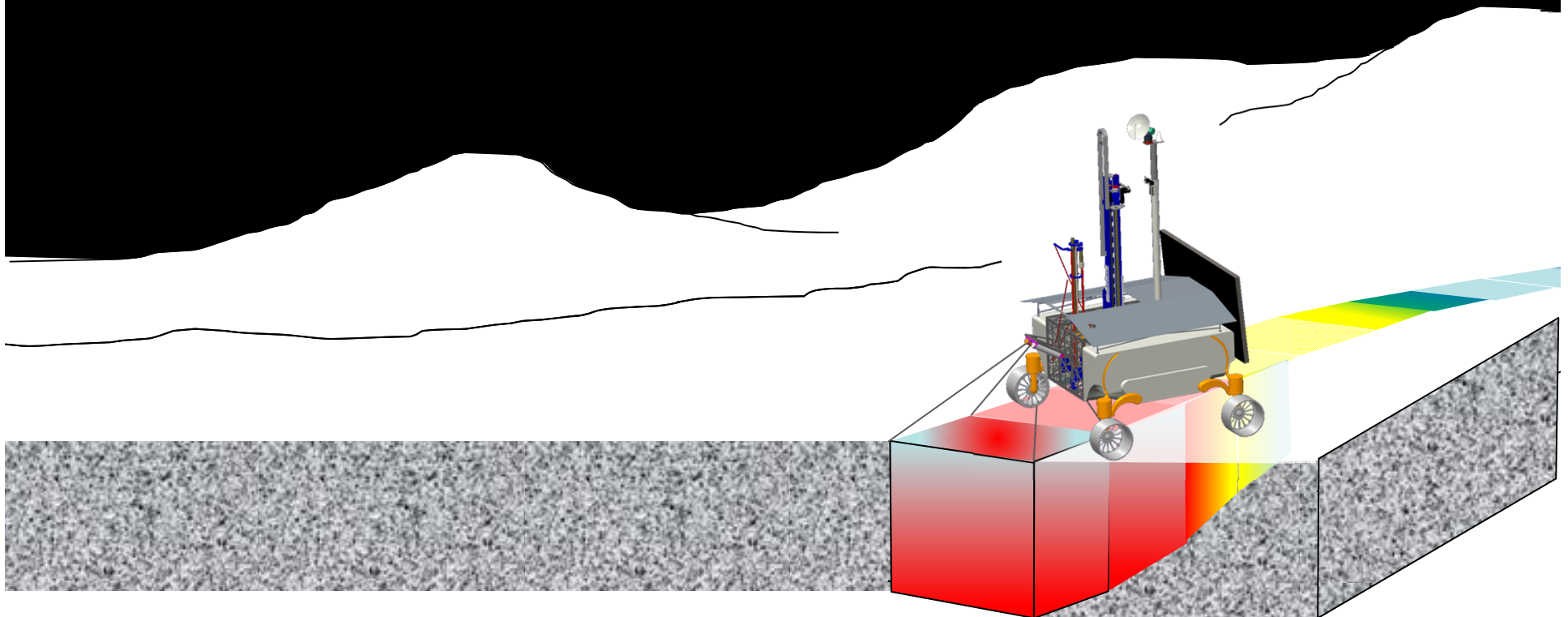




# Prospecting...



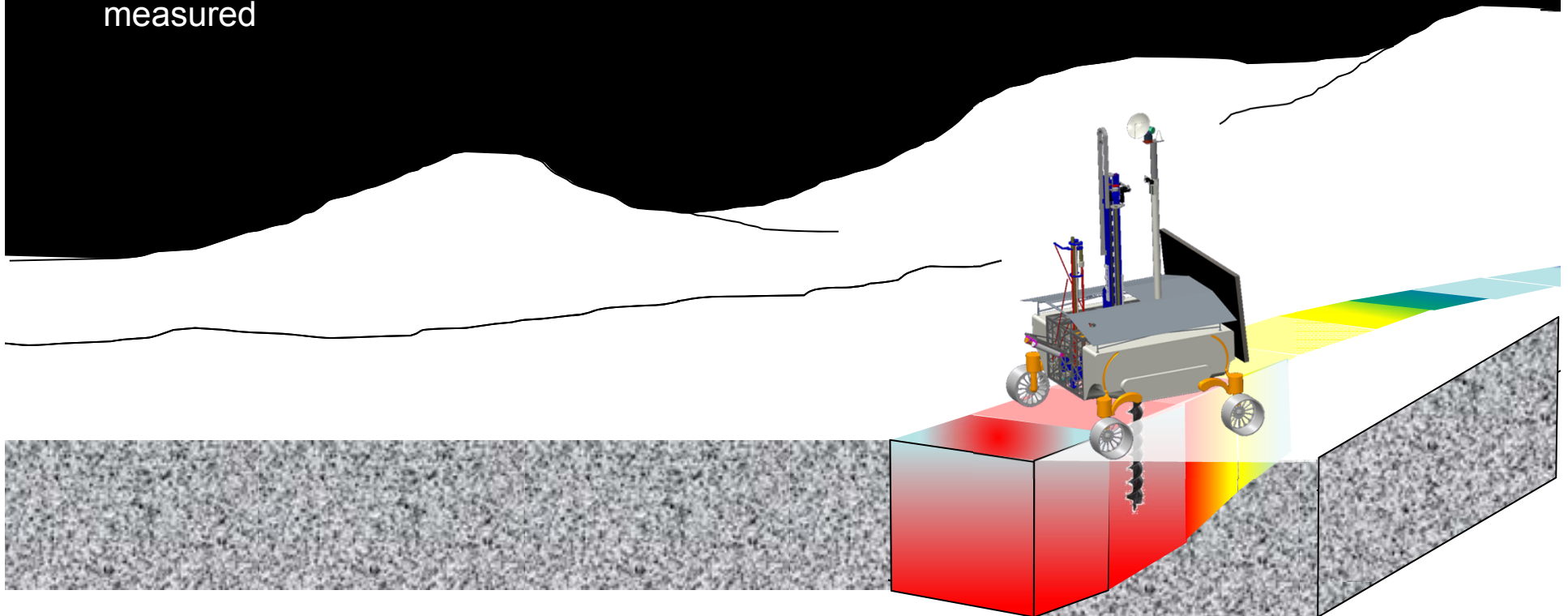
1. While roving, prospecting instruments search for enhanced surface  $\text{H}_2\text{O}/\text{OH}$  and volumetric hydrogen
2. When enhancements are found decision made to either auger or core (sample)



# Excavating...



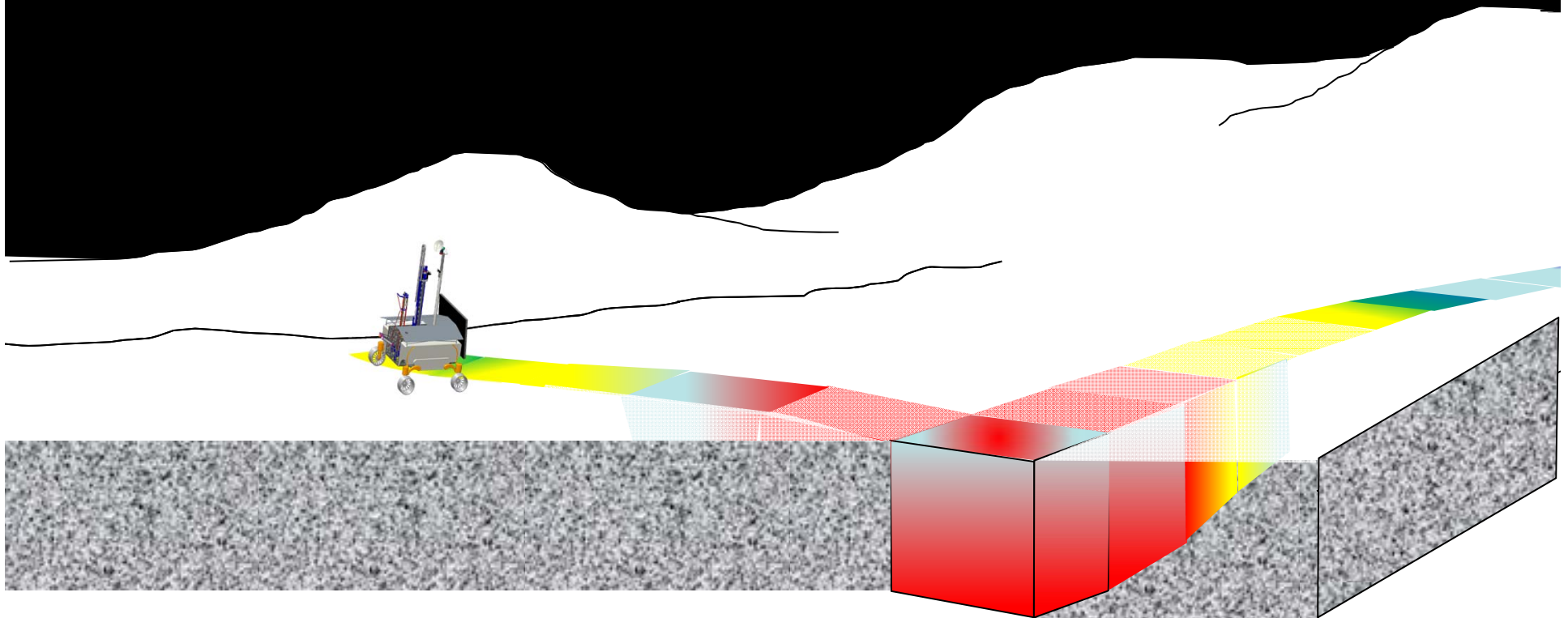
1. While roving, prospecting instruments search for enhanced surface  $\text{H}_2\text{O}/\text{OH}$  and volumetric hydrogen
2. When enhancements are found decision made to either auger or core (sample)
3. Samples are processed and evolved volatiles measured



# Mapping...



Mapping of volatiles and samples continue across a variety environments, testing theories of emplacement and retention, and constraining economics of extraction.





# Resource Prospector – The Tool Box



## The Rover

- 300 kg class rover
  - Solar powered, DTE comm
  - Mobility system
- Active suspension
- All-Wheel Steering & All-Wheel Drive
- Cameras
  - Navigation & Hazard
  - Hazard Detection / "Virtual Bumper"
- Nominal speed is 10 cm/s (Prospecting) with sprint speeds of 50 cm/s



# Resource Prospector – The Tool Box

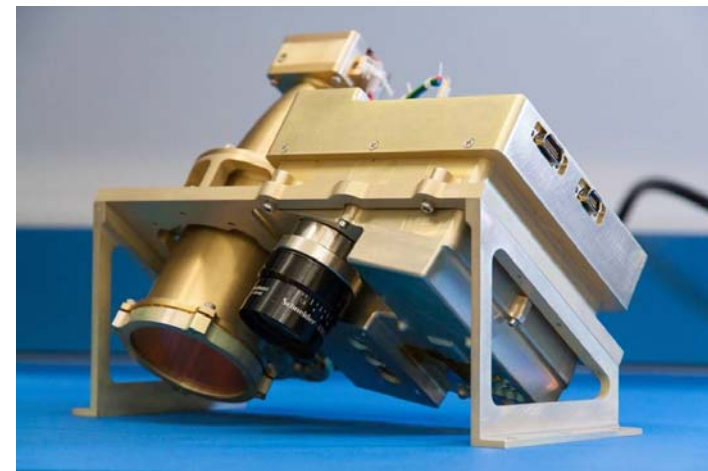
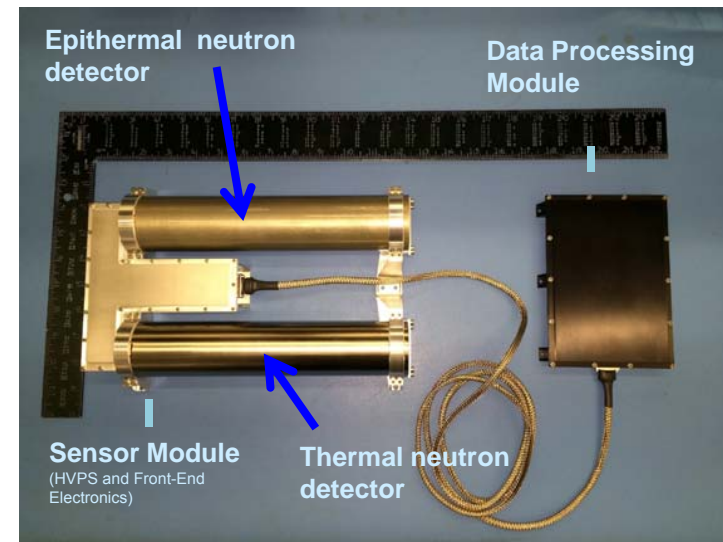


## Neutron Spectrometer System (NSS)

- Water-equivalent hydrogen > 0.5 wt% down to 1 meter depth with 2 m spatial sampling

## NIR Volatiles Spectrometer System (NIRVSS)

- Surface H<sub>2</sub>O/OH identification (1.6-3.4  $\mu\text{m}$ )
- Subsurface sample characterization (with drill)
- Multi-color imaging of drill cuttings/surface (eight colors between 0.4-1.1  $\mu\text{m}$ )
- Scene thermal radiometry (8, 10, 12 & 25  $\mu\text{m}$ )



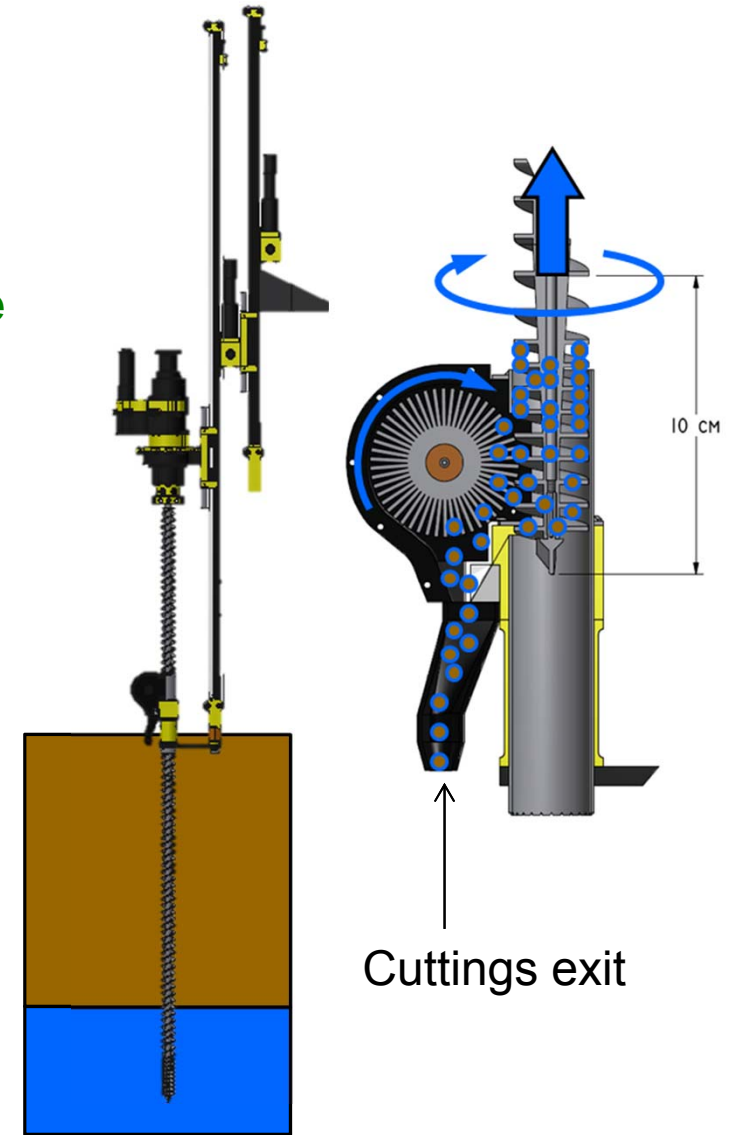


# Resource Prospector – The Tool Box



## Drill

- Subsurface sample acquisition down to 1 meter in 0.1 m “bites”
- Auger for fast subsurface assay with NIRVSS
- Sample transfer to OVEN for detailed subsurface assay





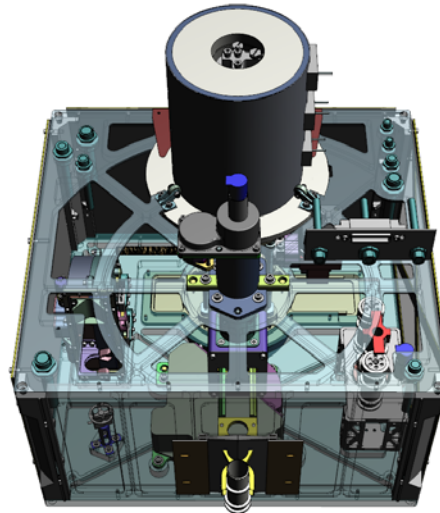
# Resource Prospector – The Tool Box



## Processing & Analysis

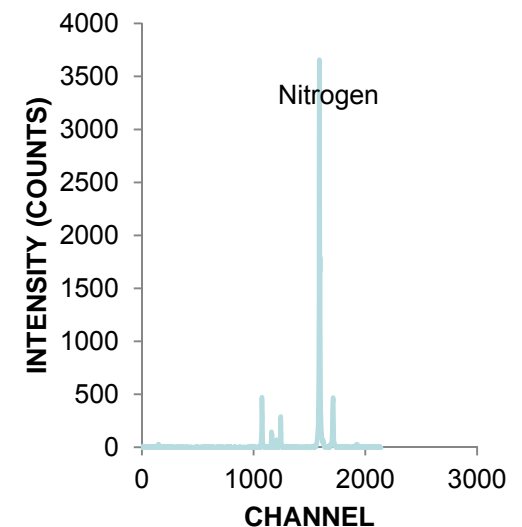
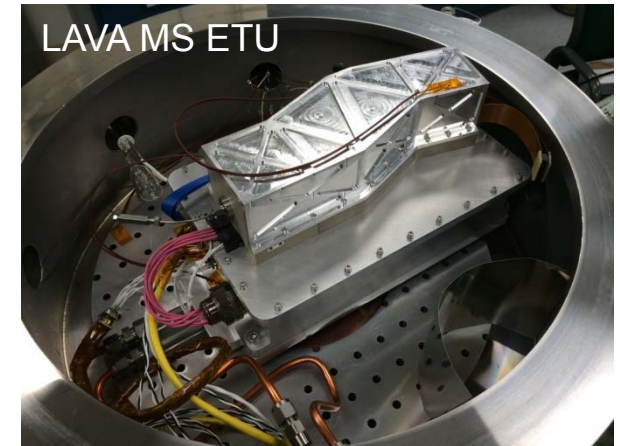
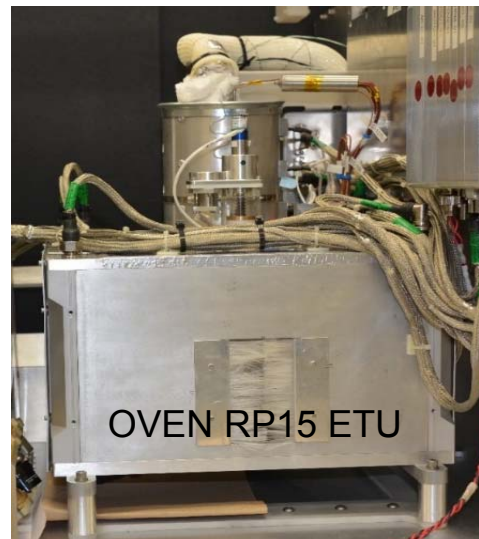
### Oxygen & Volatile Extraction Node (OVEN)

- Volatile Content/Oxygen Extraction by step-wise sample heating (150 to 450C)
- Total sample volume & mass



### Lunar Advanced Volatile Analysis (LAVA)

- Analytical volatile identification and quantification in delivered sample with GC/MS
- Measure water content of regolith at 0.5% (weight) or greater
- Characterize volatiles of interest below 70 AMU



Mass spectrum of air measured using TRL5 mass spectrometer system.

# RP Landing Site Selection



A good bit of the landing site requirements are driven by L1.1:

## 1.1 RESOURCE PROSPECTOR SHALL LAND AT A LUNAR POLAR REGION TO ENABLE PROSPECTING FOR VOLATILES

Full Success Criteria: Land at a **polar location** that maximizes the combined potential for obtaining a **high volatile (hydrogen)** concentration signature and **mission duration** within **traverse capabilities**



# RP Landing Site Selection



## Polar landing site based on meeting the following four criteria

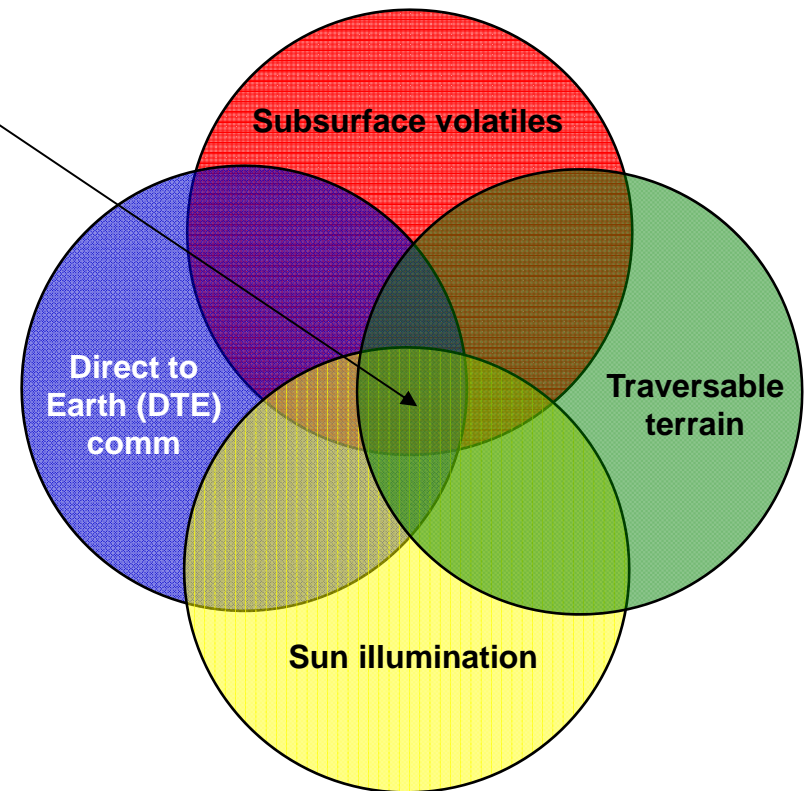
### 1. Surface/Subsurface Volatiles

- High hydrogen content (LRO LEND instrument)
- Constant  $<100$  K temperatures 10 cm below surface (LRO Diviner instrument)
- Surface OH/H<sub>2</sub>O (M<sup>3</sup>, LRO LAMP & Diviner)

### 2. Reasonable terrain for traverse

### 3. Direct view to Earth for communication

### 4. Sunlight for duration of mission for power



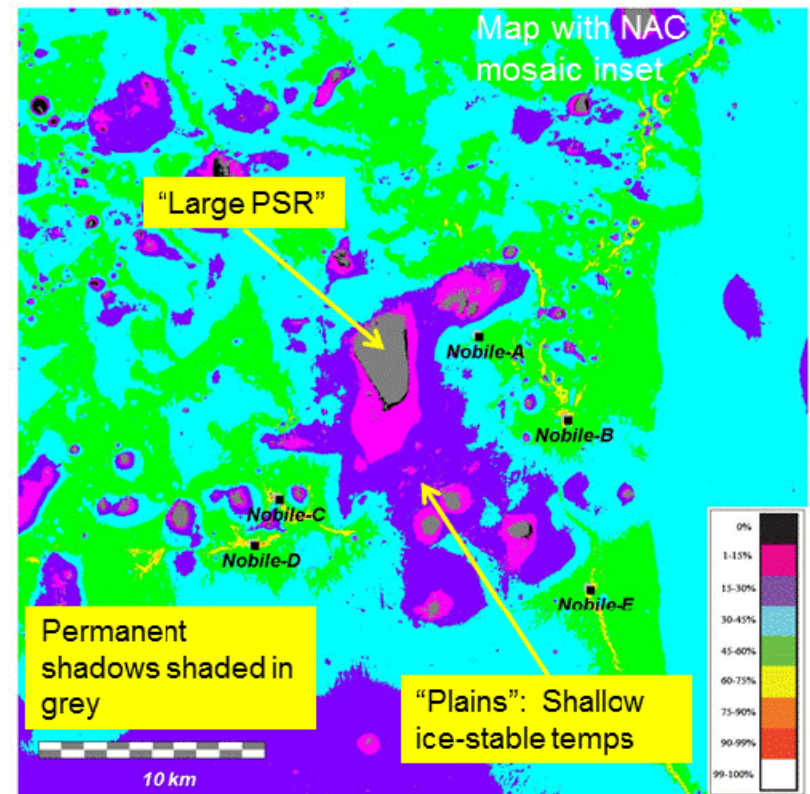


# Site Analysis



- Have considered several sites in detail
  - SP and NP
- Considered all criteria to the limits of existing data and data products
  - All meet RPM requirements
  - Preliminary hazard analysis for landing completed
- Working with UCM and APL on long-duration traverse planning
  - Utilizes corridors of connected sun and DET comm
  - From persistently-lit region traverses of 10s of kilometers, over several weeks may be possible

## North Nobile Study Site



# Summary



- RP will address a multitude of HEOMD SKGs with a robust payload
- A landing site selection process has been developed and a number of sites shown to meet mission requirements (looking toward a site analysis workshop next summer)
- Mission operations and traverse planning and process, have been developed and show mission goals can be met within power, time, etc.
- In FY15, RP is developing (building) a rover prototype with Payload to demonstrate the fundamental aspects of the mission





Thank you  
from,

*Resource*  
**Prospector**



# Extensibility to Mars and Small Bodies



SKG		Gap-Filling Activity	RP* Relevance / Feed-forward
Mars			
B4	Dust Effects on Surface Systems	B4-2. Dust physical, chemical and electrical properties	
		B4-3. Regolith physical properties and structure	
D1	Water Resources	D1-2. Water ISRU demo	
		D1-3. Hydrated mineral compositions	
		D1-4. Hydrated mineral occurrences	
		D1-5. Shallow water ice composition and properties	
		D1-6. Shallow water ice occurrences	
Small Bodies (NEAs, Phobos & Deimos)			
II. Understand how to work on or interact with the SB surface.	B. Hazards to equipment and mitigation	II-B-1. Mechanical/electrical effects of SB surface particles.	
	C. SB surface mechanical properties	II-C-2. Geotechnical properties of SB surface materials.	
IV. Understand the SB resource potential.	A. NEO resources	IV-A-4. Prepositioning and caching extracted resources.	
		IV-A-2. Knowledge of how to excavate/collect NEO material to be processed.	
	B. Phobos/Deimos resources	IV-B-1. Phobos/Deimos subsurface resource potential.	
		IV-B-2. Knowledge of how to access resource material at depth	

# Mission Goals & Relevance



## Resource Prospector: A mission to explore lunar polar volatiles

### Prospecting Mission:

- Characterize the distribution of water and other volatiles at the lunar poles
  - Map the surface and subsurface distribution of hydrogen rich materials
  - Determine the constituents and quantities of the volatiles extracted
    - Quantify important volatiles:  $\text{H}_2$ , He, CO,  $\text{CO}_2$ ,  $\text{CH}_4$ ,  $\text{H}_2\text{O}$ ,  $\text{N}_2$ ,  $\text{NH}_3$ ,  $\text{H}_2\text{S}$ ,  $\text{SO}_2$
  - Measure or provide limits on key isotope ratios, including D/H,  $\text{O}18/\text{O}16$ ,  $\text{S}36/\text{S}34$ ,  $\text{C}13/\text{C}12$

### ISRU Processing Demonstration Mission:

- Demonstrate the Hydrogen Reduction process to extract oxygen from lunar regolith
  - Demonstrate the hardware (e.g., oven, seals, valves) in lunar setting
  - Capture, quantify, and display the water generated

